# A-6 Intruder WALK AROUND



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**A-6** Intruder

By Lou Drendel





Walk Around Number 2 squadron/signal publications

## Introduction

During my research for this book, I was hanging around the VA-65 maintenance spaces at NAS Oceana when one of the Senior Chiefs came up with a quote, which says more about (a) the remarkable A-6 Intruder and (b) the current state of the U.S. military, than the thousands of words which have been written about both subjects. He stated, "I have been working on this airplane half my life, and now we are getting rid of it, just when it is better than it has ever been!" The Chief was about thirty-five years old, so it is doubly ironic to consider that the A-6 Intruder was born about the same time as he was!

The extraordinary career of the A-6 has spanned years which saw an explosion of technology, from vacuum tubes to solid state integrated circuits, from low frequency radio ranges to satellite global positioning systems, from dumb bombs to smart bombs, and finally, to plastic wings! As awkward and ungainly as it looks, the airframe of the Intruder has accommodated all of these changes with equal aplomb, proving conclusively that "Pretty is as Pretty does". The fact is that the Intruder is more mission-capable today than it has ever been, which is testament to the genius of it's designers. The Intruder is also a member of another very exclusive club. It is one of a very few U.S. military aircraft which have never been exported or sold to a foreign power. It has only been flown operationally by the U.S. Navy and U.S. Marine Corps. It has distinguished itself in combat in Vietnam, the Gulf of Sidra and the Persian Gulf. It stood the cold war watches which ultimately led to victory in that long and costly conflict.

Sadly, by the time you read this the A-6 Intruder will be gone. The cost-cutters have come to Washington, and the military is suffering at the hands of those who do not subscribe to the doctrine that preparedness is the ultimate deterrent. There is no dedicated replacement for the

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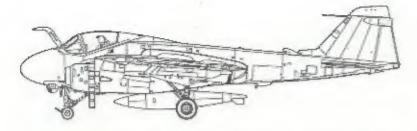
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Intruder. It's mission will be assumed by a combination of FA-18s and F-14 "Bombcats". As sleek and supersonic as they are, they will not match the load-carrying capability and deadly precision of Grumman's premier anacker.





# Grumman A-6E (TRAM) Intruder





The final attack variant of the intruder was the A-SE TRAM. This A-SE of VA-42 was on final approach to NAS El Centro, California on 15 January 1994. VA-42 remained the East Coast crew training squadron throughout the operational life of the A-6.

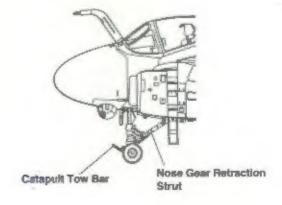




The nose landing gear strut has the estapult tow bar attached to the front. The large circular object attached to the right side of the strut is the hydraulic nose wheel steering mechanism.



The nose geer door has the landing light mounted in the center of the door, along side this light are the three carrier landing algorithms.





The port side of the nose wheel door has the carrier landing system lights. The LSO is able to tell if the aircraft is fast (Green, top light) on speed (ember, middle light) or slow (red, bottom, light) as it approaches the carrier deck for an arrested landing.

This A-8E is holeted on jacks for landing gear maintenance. The stabilizing oleo scissors link is fully extended. The Black lines running down the retraction strut are nose wheel steering hydraulic lines.





The catapult tow bar is in the raised position. When the aircraft is positioned on the catapult the bar drops down and engages the catapult shuttle. Once the aircraft is released from the catapult, the bar returns to the raised position.



There are only very slight differences between the nose landing gear of the A-6A through A-6E variants and the nose landing gear of this EA-6B Prowler.



The nose gear doors fit anugly against the bottom of the air intakes on the A-6E. The large light in the center of the small nose gear door is the landing light.

The nose landing gear well of an A-6E. The interior color is Gloss White, which is used as a maintenance tool, because things like hydraulic fluid leaks will be very visible against it. The edges of the gear doors are Red.





The TRAM turnet sensor windows are, left to right; laser designator, Forward Looking Infrared (FLIR), and laser receiver.



This is the stowed position of the A-SE TRAM turret. When in the stowed position the turret is rotated 180 degrees so that the sensor windows are protected from damage while the aircraft is parked.





The laser receiver lens (right) is more visible in this view of the TRAM turret. The large lens is the Forward Looking Infrared (FLIR) camera system. The turret can rotate 360 degrees, and can also pivot downward to track targets as they get closer to the aircraft.



The turret system allows the sensors to remain locked onto a target as the A-6 maneuvers to drop bombs. The TRAM turret can be slaved to the radar crossheirs on the BN's multi-mode radar set.

TRAM is an acronym representing Target Recognition and Attack Multi-sensor. The turret itself is twenty inches in diameter. TRAM was introduced to the A-6 fleet during 1979.





The radome has been swung open on this A-6E TRAM for servicing of the AWAPQ-156 multi-mode radar. The Interior of the radar bay is White, while most of the "black boxes" are in fact, Gray.



The svionics equipment trays swing out to allow easy access to black boxes on both sides of the reduce.

This is the same avionics equipment tray as above, but viewed from the rear.





With the radome raised, the avionics tray can be folded out for easy access. There is another fold out avionics tray on the other side of the radar bay. The interior of the radar bay is in Gloss White, while most of the avionics boxes are Gray



At least four wiring bundles are visible when viewing the avionics tray from the side. The rack holding the Gray avionics boxes is Natural Metal.



An A-6A intruder of the Marine intruder training squadron, VMAT(AW)-202, being prepared for a mission from MCAS Cherry Point, North Carolina on 20 July 1972. The aircraft has Red Foreign Object Damage (FOD) screens over the intakes and starter carts hooked up.

A-6As and A-6Es were painted with Gloss Gull Gray uppersurieces over Gloss White undersurfaces during the 1960s and 1970s. VMAT(AW)-202 was disestablished on 30 September 1986.





The open redome reveals the large multi-mode radar scanner and smaller intermediate frequency receiver (long "egg crate") under the scanner.



A KA-6D tanker at the moment of launch from the bow catapult aboard USS AMERICA during November of 1985. The nose gear catapult tow ber has dropped down and engaged the catapult shuttle.

Port engine air intake on an A-6E. The rectangular plate in front of the intake is the boundary layer splitter plate.





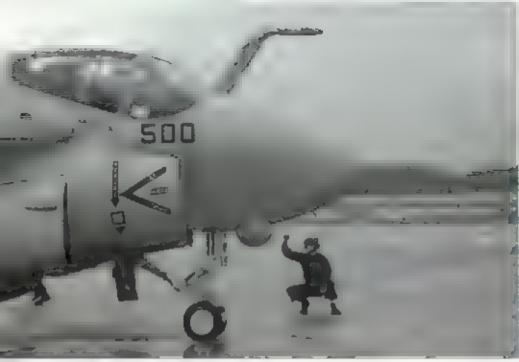
The external power receptacle is on the starboard side of the A-6, just behind the crew boarding ladder.



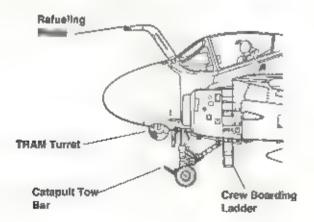
The canopy actuating mechanism is located behind the open access panel to the right of the crew boarding ladder compart-

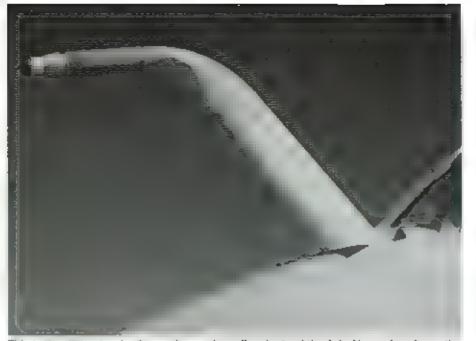


The external refueling panel is located in the same compartment as the external electrical power receptacle. The refueling of all A-6 tanks, including external tanks, is controlled from this panel.



Part of the pre-fight check is the cycling of all control surfaces and nose gear extension/ depression. Signals for all movements are given to the pilot by the Plane Captain, who monitors all movements and indicates with a "thumbs up" signal when all are operating properly.





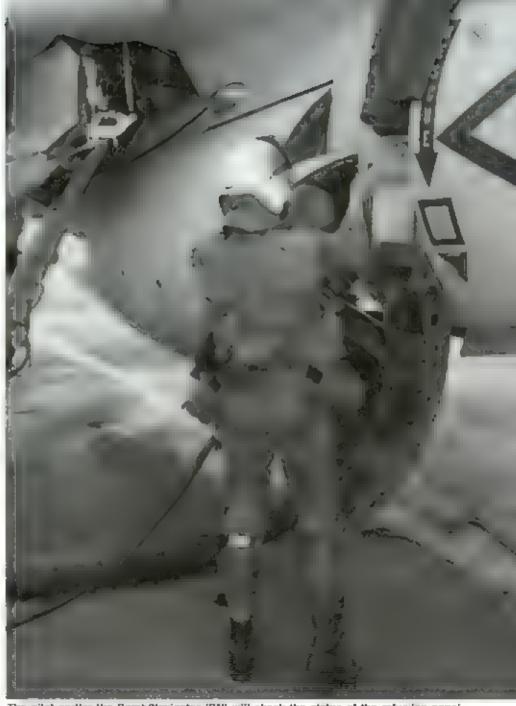
This is the air-to-air refueling probe used on all variants of the A-8. Nava, aircraft use the probe and drogue method of serial refueling, in contrast to the Air Force which uses the boom method of refueling.

The A-5 has no internal APU or method of self-starting the engines. External air is necessary and it is the responsibility of the Plane Captain to connect the starter cart hose to the engine receptacle.





The engines can not be started simultaneously, but either engine can started first. Once the engines are running, one of several ground crewmen who assist in the pre-flight will disconnect the external air hose and secure the access panels on both sides of the fuse-lage.



The pilot and/or the Bomb/Navigator (BN) will check the status of the refueling panel switches during their pre-flight walk around.



Small hydraulic jacks are used to support the nose of the A-6 when undergoing maintenance to the nose gear and related systems.



The starboard boarding ladder is used by the Bombadier/Navigator (BN). The underwing panel has been removed to service the aircraft's air conditioning unit. The large air intake at the wing root supplies ambient air for the environmental control unit.

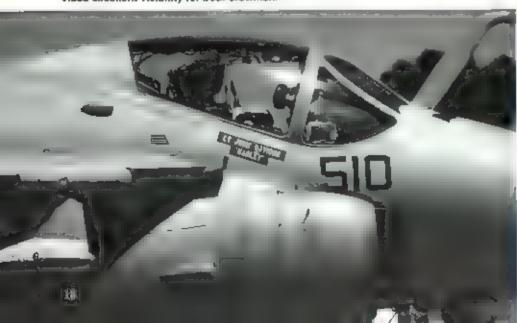


A XA-5D intruder tanker is directed to the bow cataputts aboard USS AMERICA (CV-66). Once in place, the cataputt tow bar will drop down and engage the cataputt shuttle, which will propel the aicraft down the cataputt track once the signal has been given to launch. The aircraft is carrying four external fuel tanks. The tanker can transfer the fuel from these tanks as well as the internal fuel tanks.



The engine air intake on this EA-6 is covered with a Gray fabric protective cover that also covers the Intake splitter plate. The Red craw boarding ladder is in the lowered position, although both cockpit canoples are closed. The engine cover has the equadron identification, VAQ-129, on it in Black against a White background.

The canopy on this A-6E (8uNo 152895, NH 510 of VA-95) is securely buttoned-up while the aircraft was on the ramp at NAS Miramar on 24 April 1993. The canopy on the A-6 series went unchanged until the advent of the four place EA-5B. The A-6 canopy provides excellent visibility for both prewmen.





This A-6E of VA-145 has a Green air intake cover with the squadron Identification in Yellow. The aircraft carries twenty-three fuel pump markings (Indicating buddy refuelings) on the engine bay in Black. The square at the bottom of the Rescus arrow is White with a Yellow border. On it are printed instructions on how to release the canopy.

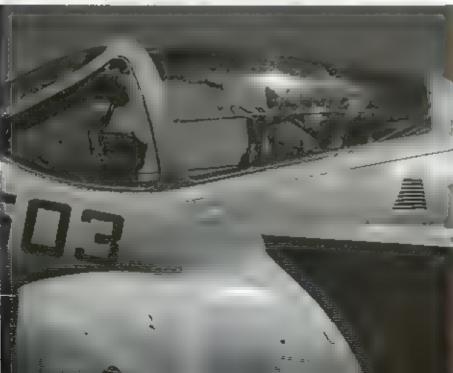
A-66 (BuNo 162211) side number AD-655 (alda number displayed as three [triple] nickels on nose) of VA-42, on approach to NAF El Centro, California on 13 March 1992. The canopy provides the pilot with excellent visibility to the side, but this wide area of glass makes him somewhat vulnerable. Flak curtains were installed on some A-5s to provide a measure of side protection for the pilot and BN.





The post antenna for the Automatic Carrier Landing System (ACLS) is located under the nose, just to the rear of the TRAM turret and offset slightly to starboard on the A-6E.

The metal bar running across the inside of the canopy of this A-6E (BuNo 181662) of VA-85 is the brace for the fisk curtain. This alreadt was one of the A-6s that took part in the Libyan raids and may have had the fisk curtains installed at that time.





Engine bleed air is used to blast rain and/or ce from the windscreen. It is directed upward from this row of vents mounted at the base of the windscreen. The small fairing just forward of these vents is the refueling light which shines on the refueling probe during night refueling operations.

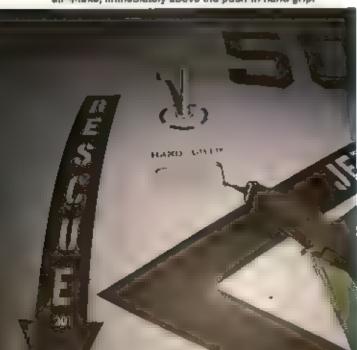
This small triangular window divides the left and right portions of the main windscreen. This window was common to all A-6 variants.





The object in the center of the danger markings on the portside air intake in the Angle of Attack (AOA) sensor.

The temperature probe is located on the starboard engine a(r-ntake, immediately above the push-in hand grip.



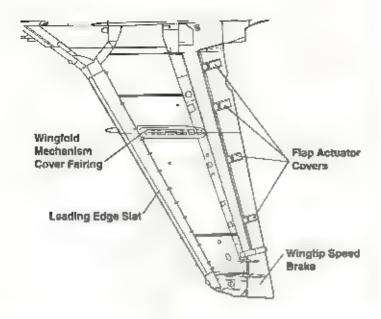


The canopy on the A-6A through A-6E stides to the rear controlled by the hydraulic piston visible at the rear of the open canopy.





All KA-6s have been retired. Those not in storage at Davis Monthan AFB are used as maintenance trainers, such as this example at NAS Fallon, BuNo 151819, formerly of VA-165. A number of interesting items are visible, including detail of the wing fold area, lowered refueling hose housing, boarding ladder and fuselage speed brake. The open panels on the fuselage are part of the engine access area.



An A-8E (BuNo 181871, WK 510) of VMA(AW)-224 in the landing patiern at MCAS El Toro, California on 26 April 1991. The aircraft has the wing leading edge stats and trailing edge full span flaps fully extended. The wingtip speed brakes are in the closed position, however when the aircraft turns onto final approach, they will be deployed.





The bulge in the center of the wing is the wingfold mechanism cover fairing. This is one of the composite-winged A-8Es.

Also visible on the wing uppersurface, besides the wingfold mechanism fairing, are the flap actuator covers and ECM fairing just behind the outboard wing fence and the fuel dump outlet, which is below the ECM fairing.





The leading edge statur the fully open position. The interior of the stat and the wing area under it are in Red. There are NO STEF legends on the stat in Medium Gray. The bulge at the top left is the wing fold mechanism cover fairing. The deriver spots on the stat are areas of fresh paint. The Gray camouflage paints used on the A-8s did not weather well and tresh paint often stood out.

The wing fold mechanism cover fairing on the older metal winged A-6s is much smaller and is not as smooth and drag tree as the new cover on the composite wing. Additionally, this wing has no fuel dump at the wingtip.





In the closed position, the leading edge slat fits anug with the top of the wing. The boundary layer wing sirilow fence helps direct the sirilow over the control surfaces at the trailing edge of the wing.

The new composite wing that two boundary layer air flow fences used to direct sirflow over the trailing edge control surfaces. Speed brake actuator fairings are outboard of the small fence at the right.





There is a speed braite mounted on each wingtip. The actuators are covered with serodynamic fairings on both the top and bottom of the speed brake. These are very effective in slowing the aircraft and are used routinely during landing approaches. Outboard of the speed brake is a low intensity strip formation light.





The wing leading edge slat in the fully entended position. The slat is controlled by two actuators visible in the White areas. Also above each actuator are White "Cautlon" warning legends.



This A-8E has the upper wing spotlers and trailing edge flaps fully extended. The spotlers are immediately forward of the full span trailing edge flaps on the uppersurface of the wing. The spotlers are used to dump lift and enhance braking effectiveness on landing rollout. The aircraft is also carrying Multiple Ejector Racks (MERs) on the inboard underwing pylons. The MERs are configured with Blue twenty-five pound practice bombs.

The Interior of the wingtip speed brake is Red with a White warning legend. Immediately to the right of the speed brake is a low intensity formation light. On the left side is the fairing for an Electronic Countermeasures (FCM) antenna. Directly under the ECM fairing is the fuel dump vent.





Environmental control system ram air intake is the larger of the two sir scoops visible on the upper rear fuseinge. The smaller scoop at the left is a cooling air intake for the avionace bay.

The emergency Ram Air Turbine (RAT) is extended from the rear of the port wing root to provide emergency electrical power in case of an engine failure.









An A-6E (BuNo 155637, NJ 811) of VA-128 on final approach for landing at MCAS El Toro, California on 22 April 1988. The wingtlp speed brakes are fully delpoyed to allow the pilot to keep the engines spooled up at higher power settings in case he needs to make a missed approach end go around. The interiors of the leading edge sist, speed brakes and flaps are in Red. The landing gear legs are tuity extended with their oleo pistons at the full travel position.



An A-6A aboard USS KITTY HAWK (CV-63) during December of 1968. The aircraft is taxling with the wings folded, a normal practice aboard ship where space is always tight. The aircraft also has ECM antennes on the leading edge of the outboard underwing pylons. The White external fuel tanks both carry the squadron number on the nose in Black.



The wing flap in the extended position. The object in the center is the flap track fairing. Full span flaps are extended to 20 degrees for both takeoff and landing.

The upper fuselage of the A-6 has several air ecoops. The large scoop is for the environmental system ram air intake, while the smaller scoop forward on the fuselage is a cooling air intake for one of the evicotics beys.



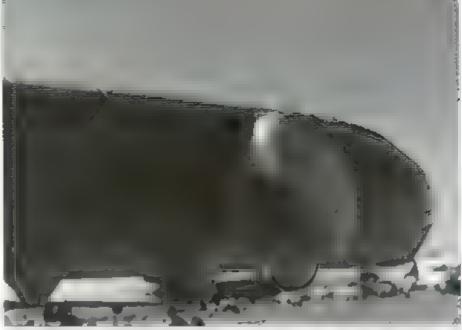


The wing flap in the retracted position. Visible are the flap track fairing (center) and boundary layer wing sirflow fence.

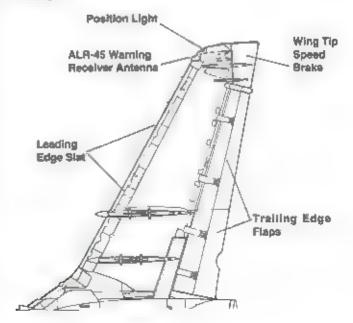
The cooling air spill vent is located under the canopy rail on the port side of the aircarit.







The large bulge under the wing tip leading edge is the entenns for the ALR-45 radar warning receiver. The fairings on the wing trailing edge are the acuators for the wing tip speed braits. Immediately next to the ALR-45 fairing is the Red part position light. There is also a small White light positioned on the wing leading edge, above and to the left of the position light.





The outboard main lending gear door actuator mechanism. The hydraulic platen on the actuator is extended. All geer door operating parts and the interior of the doors themselves are painted Gloss White to show fluid leaks. The door edges are in Red.

Visible at the forward edge of the landing gost well is the sir conditioning system exhaust fairing, which is stainless steel. The fairing is made of steel because it can get extremely hot.

The outboard main landing gear door (port side) viewed from above. One of the required items to check on a preflight check is the condition of the hydraulic lines running to the door actuators.









The fairing over the wing leading edge above the main landing gear well contains a communications enterna.

The interior of the main landing gear wheel wall is a mass of hydraulic, electrical, pneumatic and other lines. The interior of the wheel wall is Gloss White.





The knife edge fairing on the wing leading edge next to the fuselage is a stall warning buffet strip. The open access panel is for the navigation computer.

When the flaps are deployed, the flap acutator on the undecade of the wing slides out of the actuator tairing.



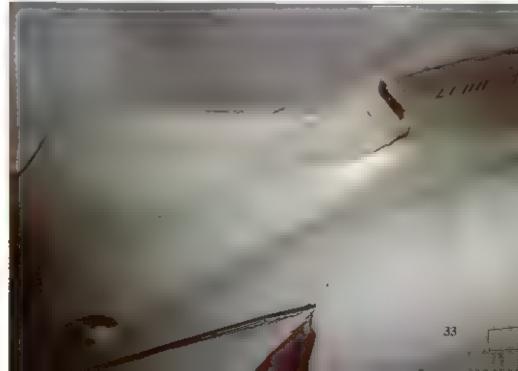


The trailing edge flaps can be extended some twenty degrees for landings and takeoffs. The object in the foreground is the flap actuator/guide. Also visible is the trailing edge of one of the underwing ordnance pylons.



Main tending gear wheel well (inboard le at top, forward is at the jeft). The landing gear retracts forward and the wheel rotates to lay flat within the wheel well.

The large intake on the starboard wing root is for the aircraft's air conditioning system. Next to it is the starboard wing stall warning buffet strip. The bright fairing at the bottom is the sir conditioning air exhaust port.





A Marine A-6E (EO 403, BuNo 161689) of VMA (AW)-533 on approach to Nellis AFB, after returning from a Red Flag mission on 23 August 1989. The aircraft is carrying Multiple Ejector Racks (MERs) on the outboard pylons and a range instrumentation pod on the starboard inboard pylon. The port MER has a Blue twenty-five pound practice bomb in place on the forward outboard rack. The landing gear on the A-6 retracts forward, with the wheel rotating to fit flat in the wheel well. The fuel tank under the fuselage is a 300 gallon tank. This same tank can be carried on the wing pylons as well.





Late production A-SE TRAMs were fitted with a regressed facing Electronic Countermeasures (ECM) antenna and fairing on each wing. The antenna was mounted at the juncture of the full span flaps and wingtip speed brakes. The fuel dump vent is located under the ECM fairing on each wing. These ECM fairings were originally developed for the A-SF program and later retrofitted to the A-SE under the Stand Off Wespons program.







The large underwing leading edge fairing is the antenna for the Radar Homing And Warning (RHAW) receiver. The fairings in the foreground are the winglip speed brake actuator fairings.

Inboard fuseings area near the main landing gear well. The large White object is the main landing gear sidestay and under the main landing gear sidestay and under the main landing gear sidestay and under the main landing gear well.





Late production A-6E TRAMs were fitted with a rearward facing Electronic Countermeasures (ECM) antenna and fairing on each wing above the fuel dump vent. These ECM fairings were originally developed for the A-6F program.

The main tanding geer well door (inboard is at the top). The door retraction mechanism is at the right. The interior of the door is in Gioss White.





The front of the port main landing gear strut, strut attachment point and retraction arm. At landing gear parts are painted Gloss White.

The main sending gear attachment point is at the left and the landing gear retraction strut is at the top.

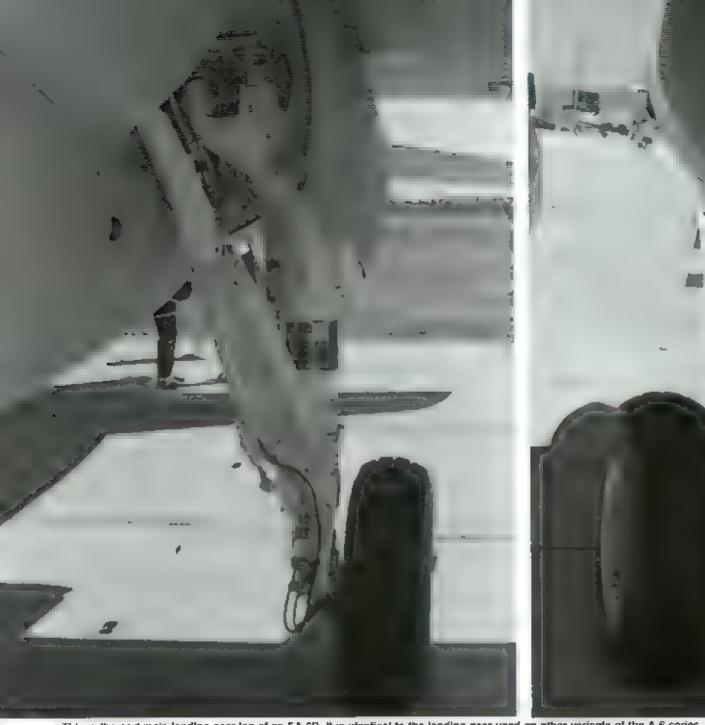


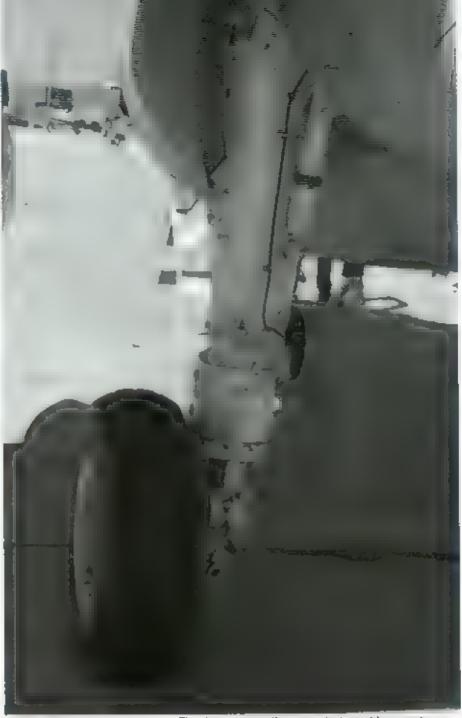


The starboard main landing gear wheel well door looking forward. The door retraction arm is in the center, along with the retraction system hydrau ic actuator.

Main landing gear strut with the aircraft tie-down chains attached to the tie down attachment points. Each wheel strut, mains and nose, have these tie down points.







This is the port main landing gear leg of an EA-6B. It is identical to the landing gear used on other variants of the A-6 series. The data plate on the main struct provides maintenance personnel the information such as the part serial number, when the struct was made, oleo pressures, and other valuable information.



This A-5 is suspended on jacks for landing gear drop checks. The Natural Metal main landing gear oleo strut is fully extended.



The wing flap cut out is to accommodate the underwing tank. It was discovered that when the flaps were lowered they struck the wing tank. The cut out prevents this from happening.

There are two main landing gest attachment points. The pivot point on the left is much larger than the one in the center. The landing gear must be made this strong since it must withstand landings at very high sink rates, as must all navel sincreft.



Looking back along the fuselage underside from the nose wheel well, there are a number of vents, engine compartment cooling air scoops, and a Red anti-collision light. The centerline pylon is configured with a 300 gallon external fuel tank.





The Inboard weapons pylon has the lower covers removed revealing the interior of the pylon. The wing pylons on the A-6 are stressed to carry up to 3,600 pounds. The two inverted V shaped objects are bomb/fuel tank sway braces, which help to hold the bomb/fuel tank steady on the rack. The plugs hanging down are electrical connectors that connect the pylon with whatever type of rack is suspended from it. The pylons are also plumbed to receive 300 gallon external fuel tanks. The field flag on the main landing gear strut is a Remove Before Flight marker for the landing gear locking pins.

This is the original wing fold mechanism used on all metal winged A-6 aircraft, including EA-6s. The wing fold joint is even with the outboard underwing weapons pylon. The wings fold hydraulically and can be folded or spread from the cockpit without outside assistance. With the wings folded, the wing span is reduced by more than half, making storage on the carrier far easier. There are four locking lugs visible on the upper portion of the told. The wing fold mechanism actuator is the V shaped object in





An A-SE TRAM (NK 500, Bullo 157000) of VA-195 at NAS Miramar, California on 5 December 1992. This alreraft was retrofitted with the new composite materials wing configuration. This retrofit was easily identified by the White cylinder in the wing fold mechanism.

The composite wingfold on an A-6E. In an attempt to prolong the service life of the A-5, a new composite material wing made of graphite, epoxy, titanium and aluminum was manufactured by the Boeing Defense and Space Group in Wichits. The original plan was for all intruders to be rewinged, but the decision to retire the A-6 from active service cut the plan just short of getting the entire fleet rawinged.





The wing fold is even with the outboard weapons pylon. This A-SE has a Penguin Mk 3 anti-ship missile hung from the pylon. The Penguin missile was developed by Norway and is used on their F-16s. It was selected by the U.S. Navy for use on the SH-608 Seahawk helicopter and is built for the Navy by Grumman. It has a range of 25 miles and can be used by fixed-wing aircraft, such as the A-S.

An A-6A (BuNo 152587) of VMA,AW)- 221 chocked and tied down on the ramp at MCAS Cherry Point, NC on 13 April 1971. The aircraft has tie down chains running from just the main landing gear struts, the nose wheel does not have tie downs attached.

The wing fold mechanism has four tooking lugs at the top of the fold. The hydrautic V shaped equator is visible in the center of the told.







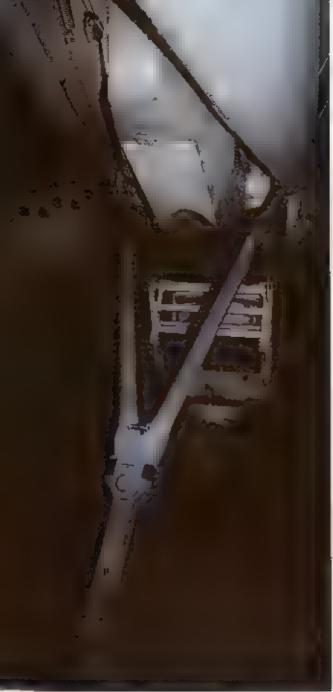
The Doppler radar radome (starboard side) and MX-7721/ALE-29A Chatt/Flare dispenser ports are located under the mid-fuselage, just shead of the tail hook.

The small under-fusalage radoms houses an APN-153 Doppter navigation radar entenna. This radar is used to provide navigational update information for the inertial navigation system...

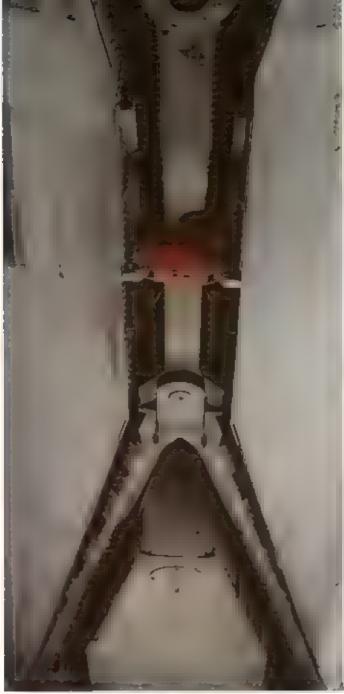




The underside of the rear fuselage with the tail hook in the stowed position. The two square ports are the MX-7721/ALE-29A Chaff /flare dispenser ports and the five points protruding downward from the fuselage are the radaz beacon antenna system.



The Y frame tailhook of a KA-6D tanker in the lowered position. The curved fairing in front of the tail hook is the hose guide for the Hose/Drum refueling unit.





The tail hook of an A-6E in the stowed (left) and lower (right) positions. The fairing in front of the A-6Es tail hook is the Doppler radar housing, which was deleted in favor of the Hose/Drum unit on the KA-6D at left. The hook is painted in Black and White stripes to make it more visible to the carrier's Landing Signal Officer



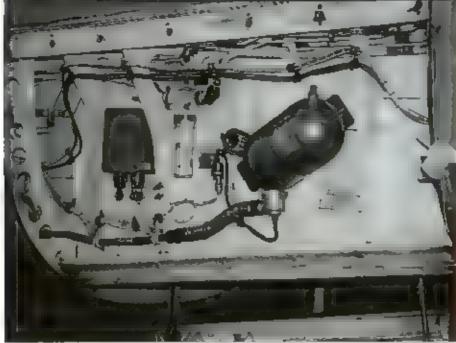
The open access panel above the lowered tail hook of this KA-60 houses Electronic Countermassines (ECM) transmitters.



The Extensible Equipment Platform on the KA-6D is hinged at the forward end. This plet form contains the refueling Hose/Drum unit. The object at the bottom is the refueling bashest that tralls below and behind the KA-6 during refueling operations.

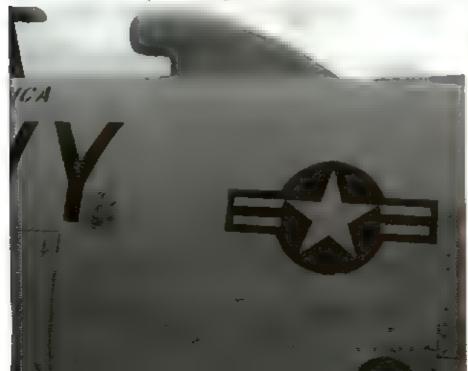


The Extensible Equipment Platform of the A-6E is hinged at the rear it holds avionics equipment, a video recorder, Chaff/Flare Dispenser controls and a maintenance ladder for servicing the Intruder's Liquid Oxygen System, which is in the fuselage just forward of the avionics bay. The mechanic holds the Green LOX bottle which will be installed before the aircraft is released for flight.



The A-6E does not have fuselage-mounted speed brakes. An equipment bay takes the place of the speed brakes mounted on early A-6As. Normally it is covered by a solid access panel.

This large ram air scoop on the A-5E is for evionics cooling in the rear equipment bays. The circle at the bottom of the "Y" is the arresting gear hydrautics reservoir sight gauge and the hole under the insignis is a cooling air exhaust vent.

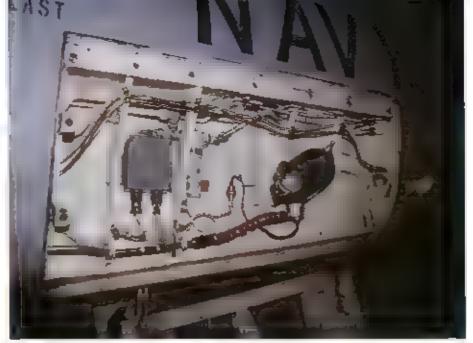




The leading edge of the A-6E fin contains the Remote Compass Transmitter as well as communications antennas and the pitot tube. The anti-collision light just under the pitot tube is Red on standard aircraft and Green on eircraft equipped with hight vision goggles.







The port side fuscinge equipment bay on an A-SE intruder The interior of the bay is Gloss White. This bay occupied the area that was taken up by the fuscinge mounted epend brakes on the A-SA. When the speed brakes were relocated to the wingtips on the A-SE, the area was then given over for equipment storage, including the fire extinguishing system supply bottle.

The starboard equipment bay has a foll covered pipe running through it along with various hydraulic and electrical lines and control valves. The Extensible Equipment Platform is visible just below the open bay.







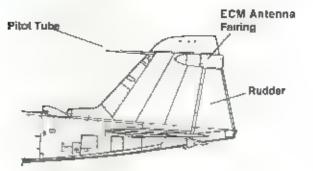
The port side fuselage equipment bay and lowered Extensible Equipment Platform on an A-6E. The legend below the fire extinguisher system bottle reads, " Service If Presure is Below 100 PSI." The Danger warning cautions against connecting electrical lines until the bottle is secure and the discharge lines are connected.



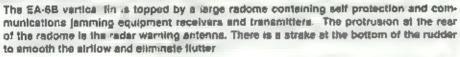
The speed brake of an KA-5D. This type of fuselage mounted speed brake was used on early A-6s, although problems with the speed brake often led to them being wired short



The fin of the A-6E contains a rearward facing Electronic Countermeasures (ECM) antenna in the streamlined fairing just above the rudder The fin cap itself contains a UHF Identification Friend Foe (IFF) antenna.



The lower portion of the extreme rear fuselage contains the rear White tall navigation light at the base of the rudder. The vent at the left is the fuselage fuel tank jettison pipe.

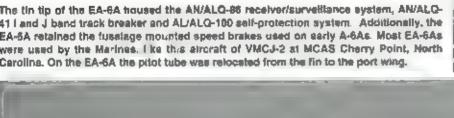






The rear fuselage, fin and tall planes of an A-6E. The fin tip houses, a rearward facing Electronic Countermeasures (ECM) antenna in the streamlined fairing just above the runder, while the fin cap itself contains a UHF Identification Friend Foe (IFF) antenna. Just below the rudder is a White position light. On the fin leading edge is a anti-collision light mounted just below the pitot tube boom. The leading edge of the fin cap has a communications antenna built into it. Just below the horizontal stabilizers is the fuselage fuel dump pipe. This is used to jettision fuel prior to landing if the aircraft is too heavy to come aboard the carzier. At the forward base of the fin is the rear fuselage

venting air Intake. The tin tip of the EA-6A housed the AN/ALQ-86 receiver/surveillance system, AN/ALQ-41 I and J band track breaker and AL/ALQ-180 self-protection system. Additionally, the EA-5A retained the fuselage mounted speed brakes used on early A-6As. Most EA-6As. were used by the Marines. I ke this sircraft of VMCJ-2 at MCAS Cherry Point, North Carolina. On the EA-6A the prior tube was relocated from the fin to the port wing.







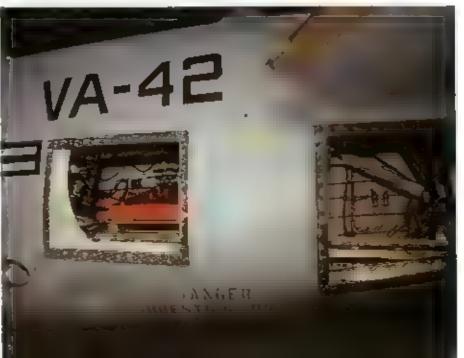


The rear fueelage fuel dump vent on the A-6E is located at the base of the fin. It allows the quick dumping of fuel so the aircraft can get down to landing weight if it returns to the carries with excess fuel. Landing weight is critical on aircraft carriers and dumping fuel is the method used to control the aircraft's weight.

This open access panel on the starboard rear fuselage reveals the stabilator control linkage on an A-6E.

These open access panels on the port rear fuselage are for (front) ECM transmitter/receivers and (rear) control linkages.







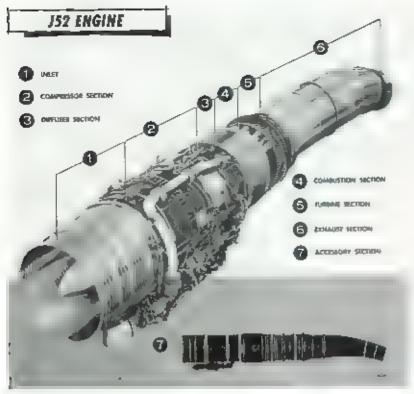
The centerline pylon on the A-6 can be used to carry ordinance or fuel tanks. Most commonly, it is configured with a 300 gailon fuel tank.



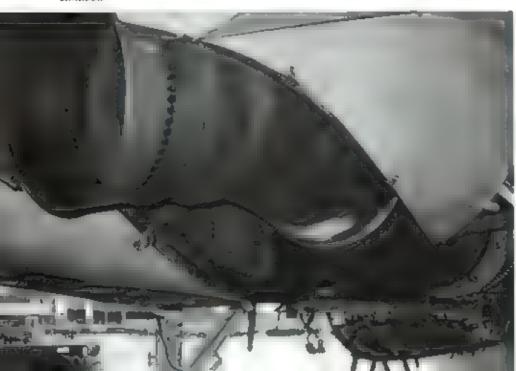
The refueling basket used on the KA-SD. The lights above the basket are the fuel flow status lights that tell the receiving pilot that fuel is flowing through the hose.

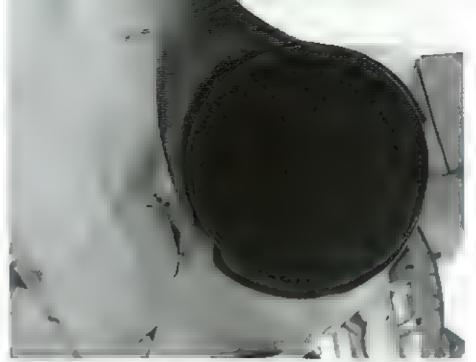
The Hose/Drum unit is mounted on the Extensible Equipment Platform and the backet is housed in a streamlined fairing when not extended. The KA-6D also retains the fuselage speed brakes.





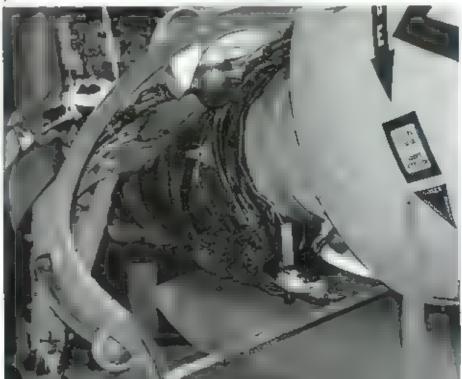
This A-6E angine has had the fuselage skinning removed for maintenance of the engine exhaust.

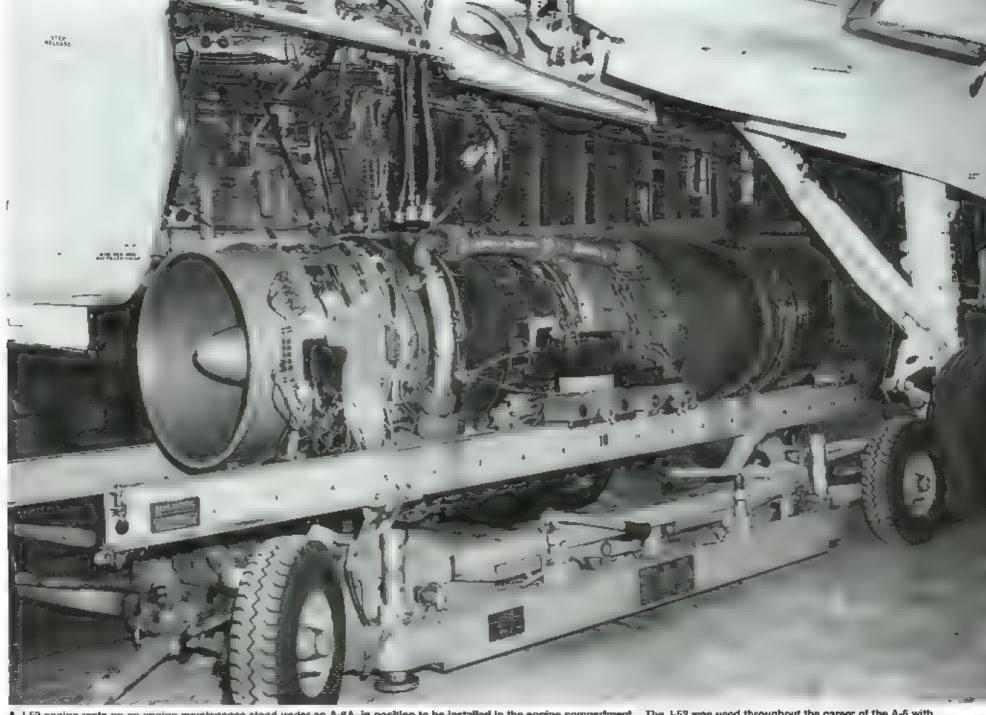




A-6E engine exhaust area. The blade antenna at the left is for the UHF communications system.

The engine compartment of an A-6E with the access panels fully open and strut locks in place.





A J-52 engine rests on an engine maintenance stand under an A-6A, in position to be installed in the engine compartment. The J-52 was used throughout the career of the A-6 with Ritle change or modification to the engine.



Engine intake of the J-52 angine. The compressor section turbine blades are visible behind the built fairing.

The port side of the J-52 engine compartment (looking rearward). The engine compartment access doors are hinged at the top.





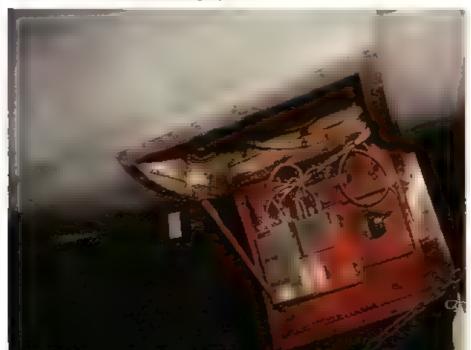
The rear fuselage (exhaust) fairing removed during maintenance. When in place, this fairing covers the engine exhaust section except for the extreme and of the exhaust pipe.

The starboard side of the J-52 engine compartment, with a J-52 installed, (looking reseward). The external power (air) receptable is at the right.





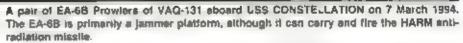
(Abova/Seiow) This open access panel reveals the radar altimeter set and its antenna (the thin water on the access panel exterior). The unit itself and various electrical connections are on the interior of the hinged panel.







The EA-6B is a four place aircraft with alda-by-side sealing for its crew. The separate cookpits each have their own canopy. The crew boarding ladder remains the same as on earlier A-5s, with a pistform to assist the front cockpit crewmen in boarding the aircraft.





The fairings on either side of the EA-6B fin are for Band 1 and Band 2 transmitter antennes. The fin cap contains self-protection and communications jamming equipment





The refueling probe of the EA-68 differs from the standard A-5 senes in that it is offset to starboard. The crewmen are seated in Martin-Balter GRUEA-7 ejection seats.



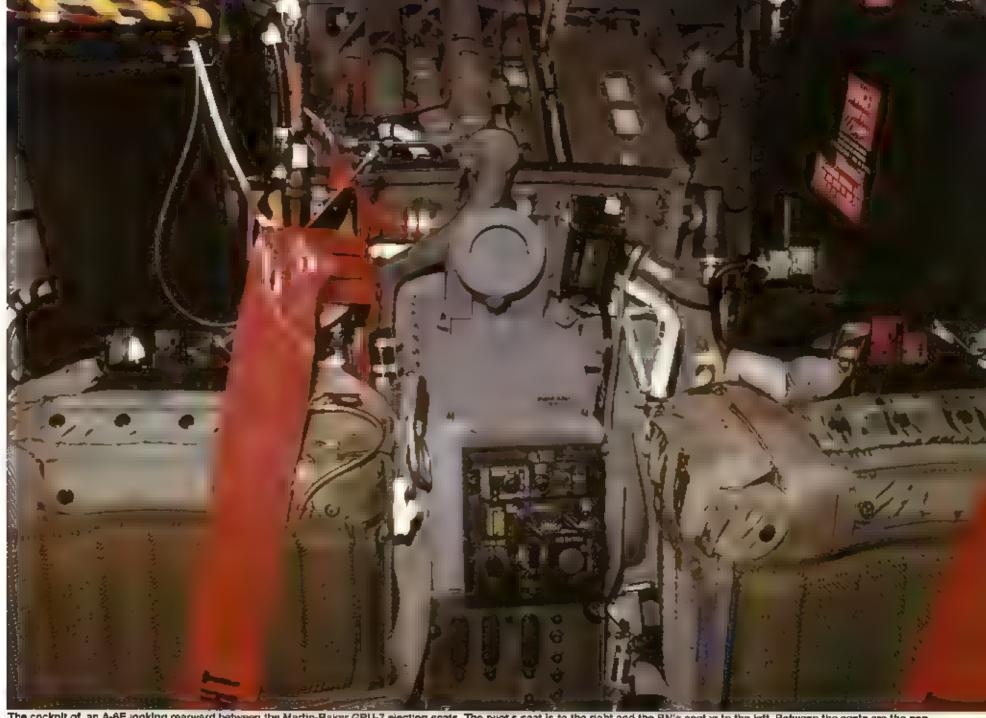
This EA-6A of VMCJ-2 at MCAS Cherry Point has the engine compartment open, the Extensible Equipment Platform lowered and the speed brakes deployed.

An EA-6A undergoing carrier qualifications. The first flight of the EA-6A was on 26 April 1963. The aircraft is configured with underwing jammer pode and three 300 gallon fuel

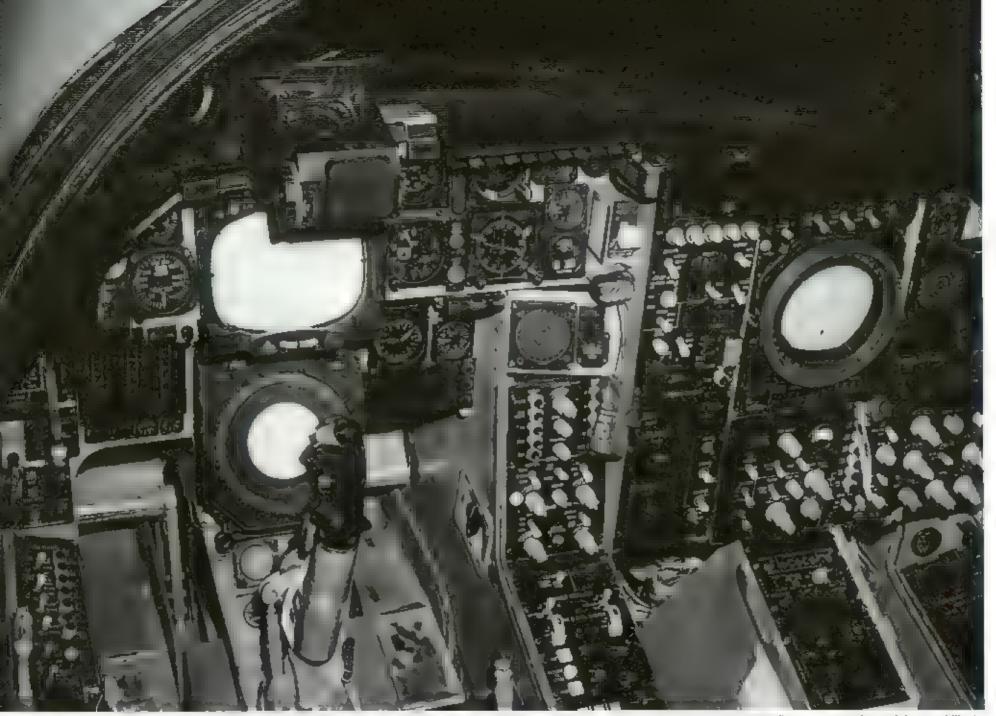




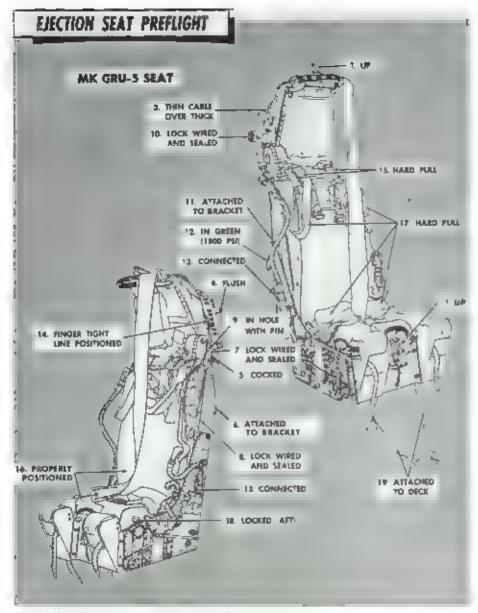
A-6E cockpit, the pilot is on the left side and the bombadier/navigator is on the right, slightly lower and further back than the pilot. The BN's cockpit is dominated by the large rader acope and tactical displays. The control stick on the BN's console is for rapid control of the weapons system. Using this stick, the BN can boresight the radar on a particular target, then slave the TRAM turret's sensors to this same traget.



The cockplt of an A-6E tooking rearward between the Martin-Baker GRU-7 ejection seats. The prior s seat is to the right and the BN's seat is to the left. Between the seats are the controls for the radiar beacon, radios, and radio compass. Both seats have "Red Remove Before Flight" banners attached to the seat safety plns.

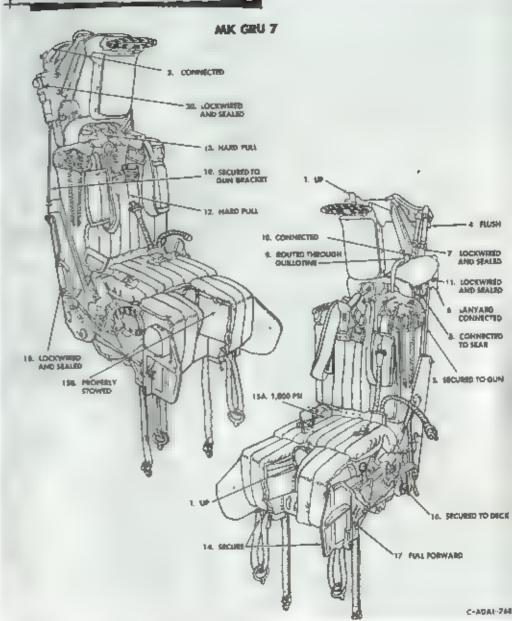


The instrument panels of an A-6B intruder. The A-6B was used for the Iron Hand (anti-SAM missile) mission and had enhanced threat identification electronics and the capability to fire the Standard and HARM anti-radiation missiles. The pilot had a radar repeater scope in front of him, as well as an optical sight mounted above the instrument panel.



Early A-6 variants used the Martin-Baker GRU-5 ejection seat.

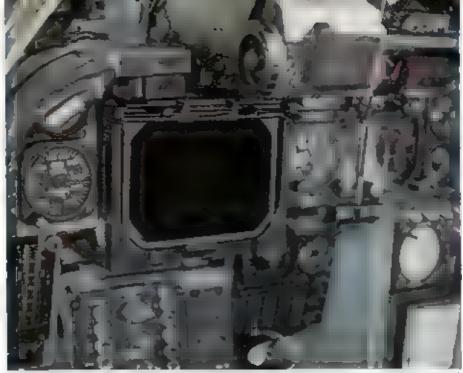
## EJECTION SEAT PREFLIGHT



Later A-6s used the Martin-Baker GRU-7 ejection seat.



The pilot's side of the A-6F cockpit. The stick is in the foreground with the thrultic quadrant to the left. The rudder pedals are on either side of the center console.



The pilot's side of the A-SE cookpit has a large vertical display indicator (CRT screen) that displays the primary flight instruments information. The radar altimeter is to the left and the barometric altimeter is on the right.

The top of the SN's seat. The Yellow/Black striped handle is the face curtain, which serves a dust purpose. When pulled out, it fires the ejection seat and it also protects the crewmen's face from wind blast.





The pilot's optical ratioctor bomb/rocket sight on an A-SE. The A-5 has no Head Up Display (HUD) for the pilot.

The "coolie hat" on top of the stick controls isteral/longitudinal trim, while the Red button next to it is the bomb "pickle". Yellow button forward of the stick grip is the autoplict emergency disengage

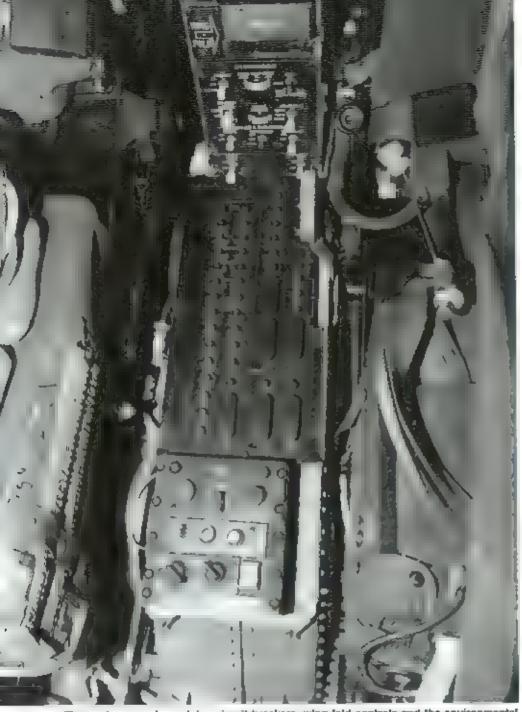




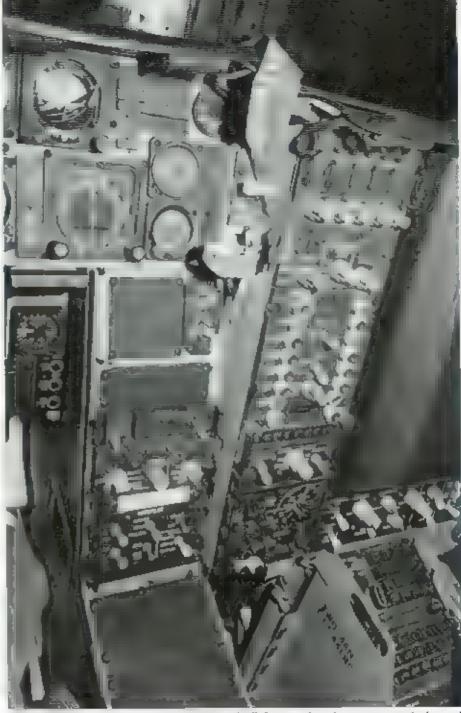
The white handle forward of the throttle quandrant folds outward and when the throttles are full forward for cataputs shot, the pilot gripe this handle in addition to the throttles to keep the force of the cat shot from sliding the throttles rearward.

The magnetic compass is mounted on the canopy center support, as far away from the other instruments as possible.

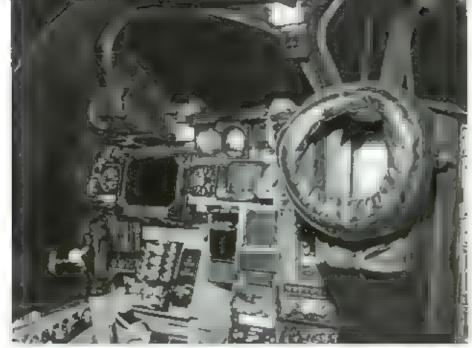




The center console contains circuit breakers, wing fold controls and the environmental system control panel (top center).

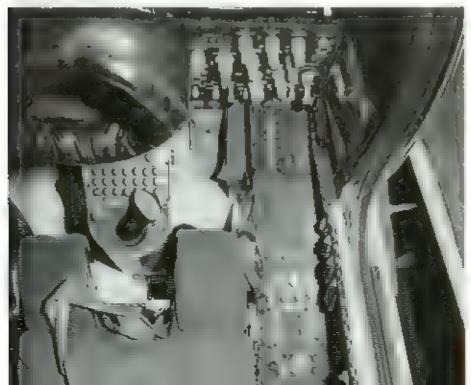


This console contains the autopilot controls (left center) and weapons control panel (right center).



The rader screen that dominates the SN's panel can be covered by a large shroud to block out outside light reflections on the screen during daylight operations.

The stick in front of the BN's seat controls the radar and weapons system. The center and right consoles contain the ECM control panel, radar test panel, Doppler control panel and other electronics.





This control panel contains the Digital Display Unit (top) and BN's master control panel and INS controls. The small gauge is the outside temperature gauge.



The canopy center framing contains lighting controls and the manual canopy release lever.



The canopy bow frame has two rear-view mirrors mounted on each side. The instructions on the canopy manual release state to "Pull Down For Access To Canopy Manual

The small lights on the canopy center framing can be trained to illuminate maps, charts, checkflats, etc.

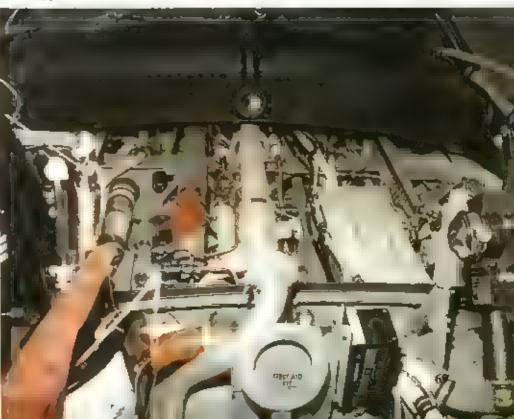






The area behind the cockpit contains electrical junction boxes, a first aid kit, electrical conduits, hydraulic lines, the canopy hydraulic jack platen housing and the weapons monitoring unit computer. The rear canopy bulkhead is painted in Flat Black to reduce reflections in the canopy.



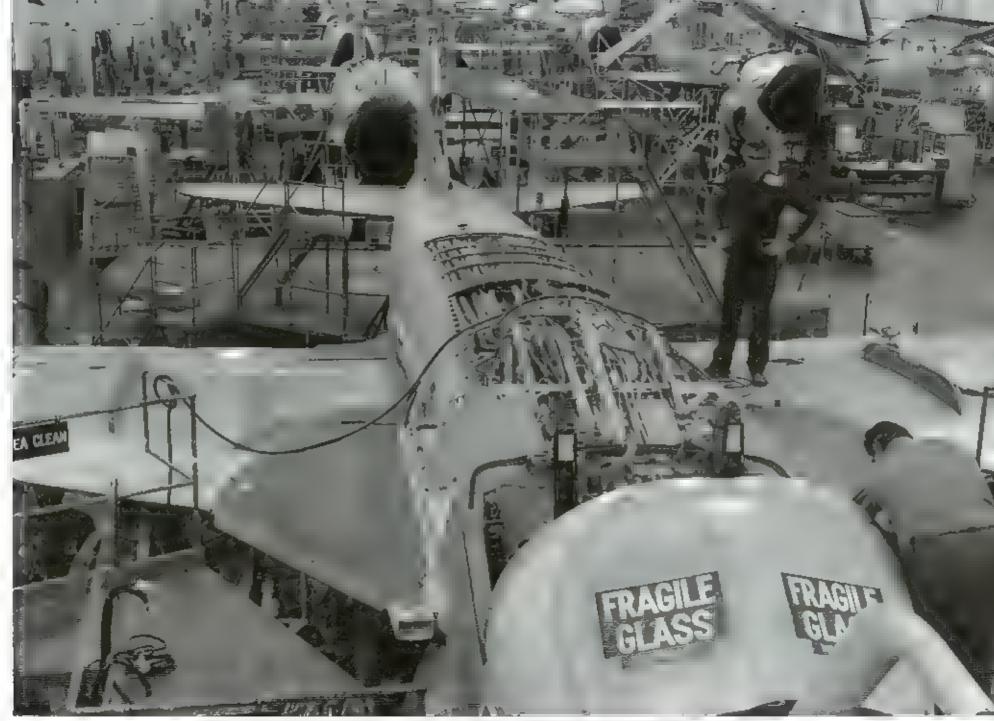




The canopy hydraulic jack actuating piston housing is in the center and the weapons monitoring computer is below it.



The canopy on the A-6 stides to the rear for entry. The separate canopies of the EA-6B raise upward. This EA-6B of VAQ-131 was aboard USS CONSTELLATION on 7 March 1994.



The A-6A assembly line in the Grumman factory at Culverton, Long Island, New York, during the early 1960s.





The open access panel on the inboard weapons pylon on an A-6E reveals the electrical connections for the pylon, wiring conduits, bomb/drop tank away braces and fasteners.

The additional fairing on the outboard side of the autboard weepons pylon covers part of the wing fold mechanism.

This is the inboard pylon on an A-5A with all panels closed and fastened down. Each pylon has two sets of sway braces to help keep whatever is hung on the pylon from shifting.



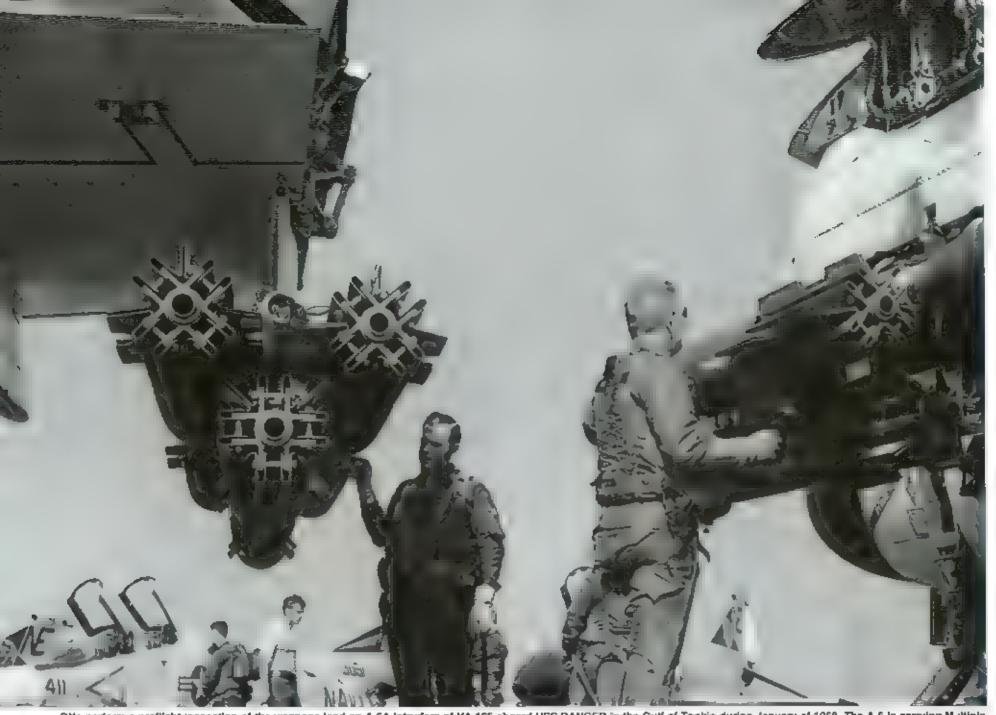


The circular object between the sway braces if the ordnance rack attachment lug. In addition to bomb racks and weapons, the pylons have the necessary plumbing for fuel tanks.

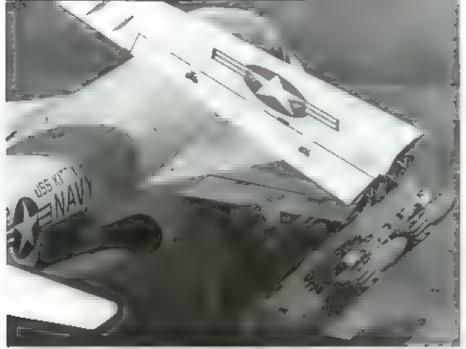


The Multiple Ejector Rack (MER) is capable of carrying six bombs, three on the forward portion and three on the rear portion. It can be carried on any of the standard A-6 weepons pylons. The suspension lugs are in the center of the rack.



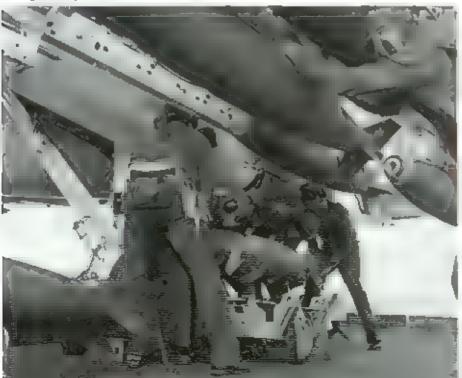


BNs perform a preflight inspection of the weapons load on A-6A intruders of VA-165 abound USS RANGER in the Gulf of Tonkin during January of 1968. The A-6 is carrying Multiple Enjector Racks (MERs) loaded with Mk 82 Snakeye high drag bombs on the outboard weapons pylons.



An A-8A of VA-85 with the wing pylons loaded with MERs carrying a lotal of twenty-four Mk 82 Snakeye bombs taxies toward the cataput aboard USS KITTY HAWK on 25 March 1957

Ordnancamen load lowdrag bombs from a bomb cart onto an A-SA aboard USS RANGER during January of 1968.





This A-SA of VA-85 aboard USS AMERICA in the Gulf of Tonkin during June of 1968 has a message chalked on one of the Mk 82s. The aircraft also has a flak curtain installed in the BN's canopy

These Red Shirts (Ordnancemen) are preparing to fuse the Mk 82s hung on the MER on this A-6A of VA-85. The 500 pound Mk 82 was one of the most widely used bombs of the Vietnam War





The only weapon cleared for the EA-68 is the HARM anti-radiation missile. This EA-68 of VMAC-2, returns to MCAS Yuma on 14 March 1992 after a captive training flight with a

This A-8E is loaded with a pair of AGM-68 HARM anti-radiation missile and twelve Mix 88 general purpose low drag bombs. The A-6E can carry a wide variety of weapons including rockets, missiles, cluster bombs, iron bombs and smart weapons.









This A-6A is loaded with two 300 gailon external fuel tanks on the inboard pylons. The A-6 can carry these tanks on all five pylons for a ferry mission.

This A-6 of VMA (AW)-224 is loaded with Mk 20 Rockeye anti-armor cluster bombs during Linebacker operations against North Vietnam on 7 May 1972.





An A-6 cryman checks the Mk 82s on his A-6 aboard USS EISENHOWER (CVN-69). The MER is loaded with only five on the inboard pylon since the sixth would not clear the landing gear door.

This was the A-6E TRAM prototype. The antennas under the nose are the VHF communications (forward), UHF communications (second long antenna) and TACAN (last short entenna).





A-6Es of VA-52 refuel from a KA-3B tanker. Prior to retirement of the KA-6D, each aquadron had four or five tanker intruders as part of their inventory. The shortfall in tanker capability was handled by the KA-3, until it too was retired. Current practice is to configure A-6 bombers with "Buddy" refueling stores on one or more stations.

An A-6E of VA-42 on final approach to NAS El Centro, California on 15 January 1994. It is configured with a "Buddy" refueling store on the centerline station.





An A-SE (VK-3 Bullo 155657) of VMA(AW)-121 at NAF Washington, during November of 1974. Except for the addition of some cooling acceps and a few new antennes, the basic shape of the intuder has not changed in thirty years.

The final intruder. The A-6F was built in prototype form only and, even though the A-12 project was cancelled, the A-6F was never funded.



